



# **NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS**



## **1/98 IMOS WORKSHOP**

**presentation by**

**Andy Kissil**

**Jet Propulsion Laboratory  
California Institute of Technology**



# **NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS**



## **Presentation Outline**

- # Overview of Using IMOS with NGST**
- # Identification of NGST Model Components**
- # Model Simplification**
- # NGST 6/97 Model Overview**
- # Model Checkout Results**
- # Thermal Distortion Analysis**
- # Using Substructuring for Very Large Models**
- # Summary of IMOS Experience with NGST**



# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## # Overview of Using IMOS with NGST (structures):

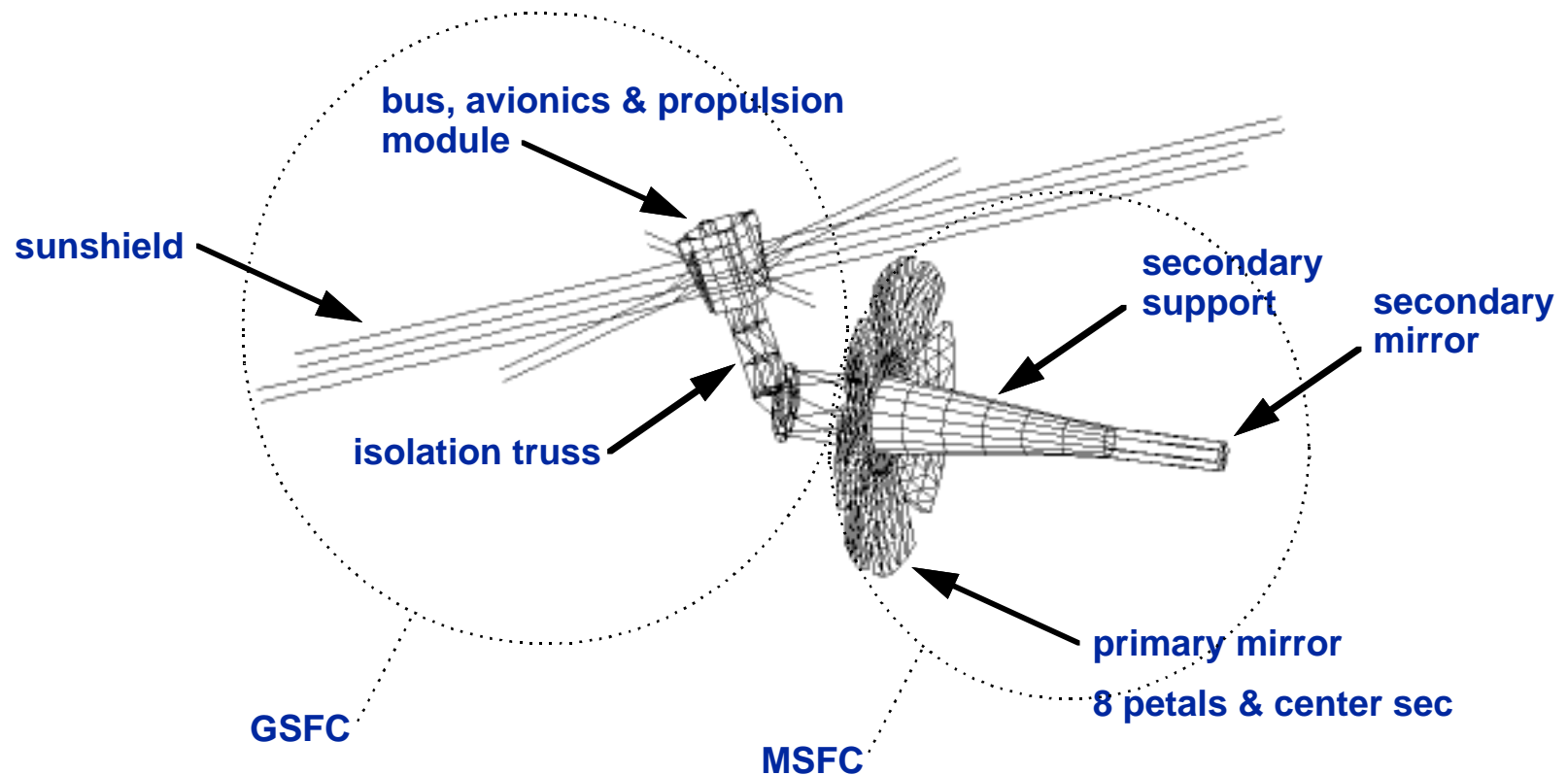
- Conversion of Integrated NGST NASTRAN Model to IMOS
  - initial model simplification of GSFC S/C structure (in NASTRAN)
  - conversion using nas2imos (generates k and m)
- Eigenvalue Analysis
  - first 100 modes (using speig)
  - animation of mode shapes (using movmak/movply)
  - mode identification using modal strain and kinetic energies
- Generation of Modal State Space Models
  - assembly of a, b, c, d Matrices (using mode2ss)
- Thermal Distortion of Telescope due to Temperature Profile (from MSFC)
  - using thermal analysis script (see viewgraph)
- Transient Time History Dynamic Simulation (fhist from GSFC)
  - using Isim
  - animation of response (using movmak/movply)
- Parametric Studies of Stiffness Characteristics



# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Identification of NGST Model Components



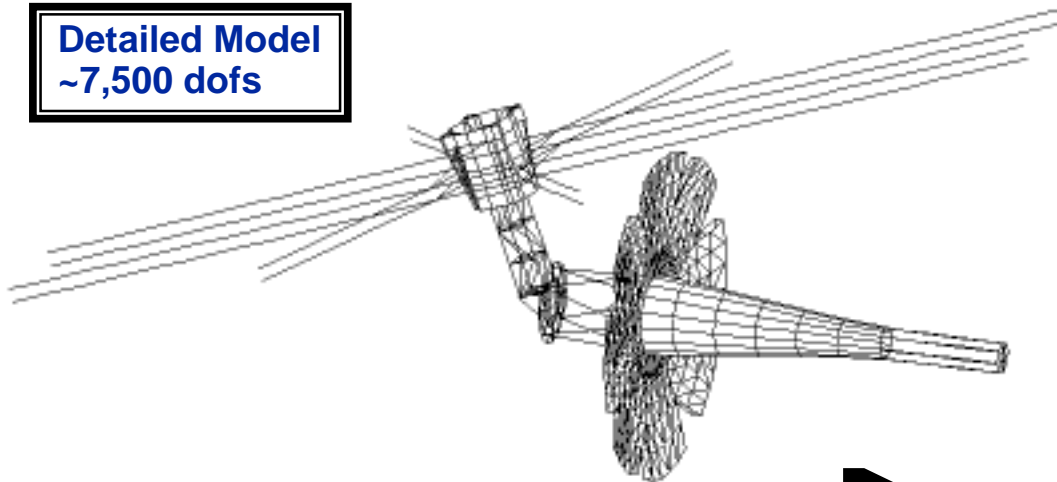


# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Model Simplification

**Detailed Model**  
~7,500 dofs



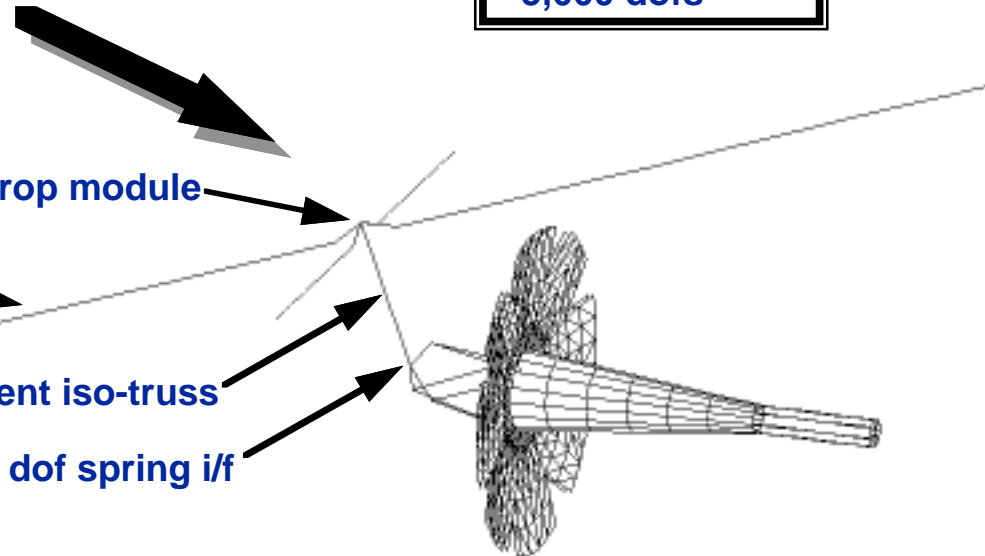
**Simplified Model**  
~5,000 dofs

6 dof pt mass bus/prop module

sunshield arms  
8-10 elts ea.

4 element iso-truss

6 dof spring i/f





# **NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS**



## **NGST 6/97 Model Overview**

- # **Nodes: 904 Total**
  - 5060 Independent Dynamic dofs
- # **Elements: 1335 Total**
  - beam 485
  - conm 134
  - celas 6
  - plate 648
  - rbe2 62
- # **Local Coordinate Systems: 12 Total**
  - 1 Spherical
  - 1 Cylindrical
  - 10 Rectangular
- # **Plate Bending and Shear Factors (and Materials) Used in NASTRAN Model**
  - Representing Honeycomb Sandwich Construction
  - Automatic Conversion Using nas2imos



# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Model Checkout Results

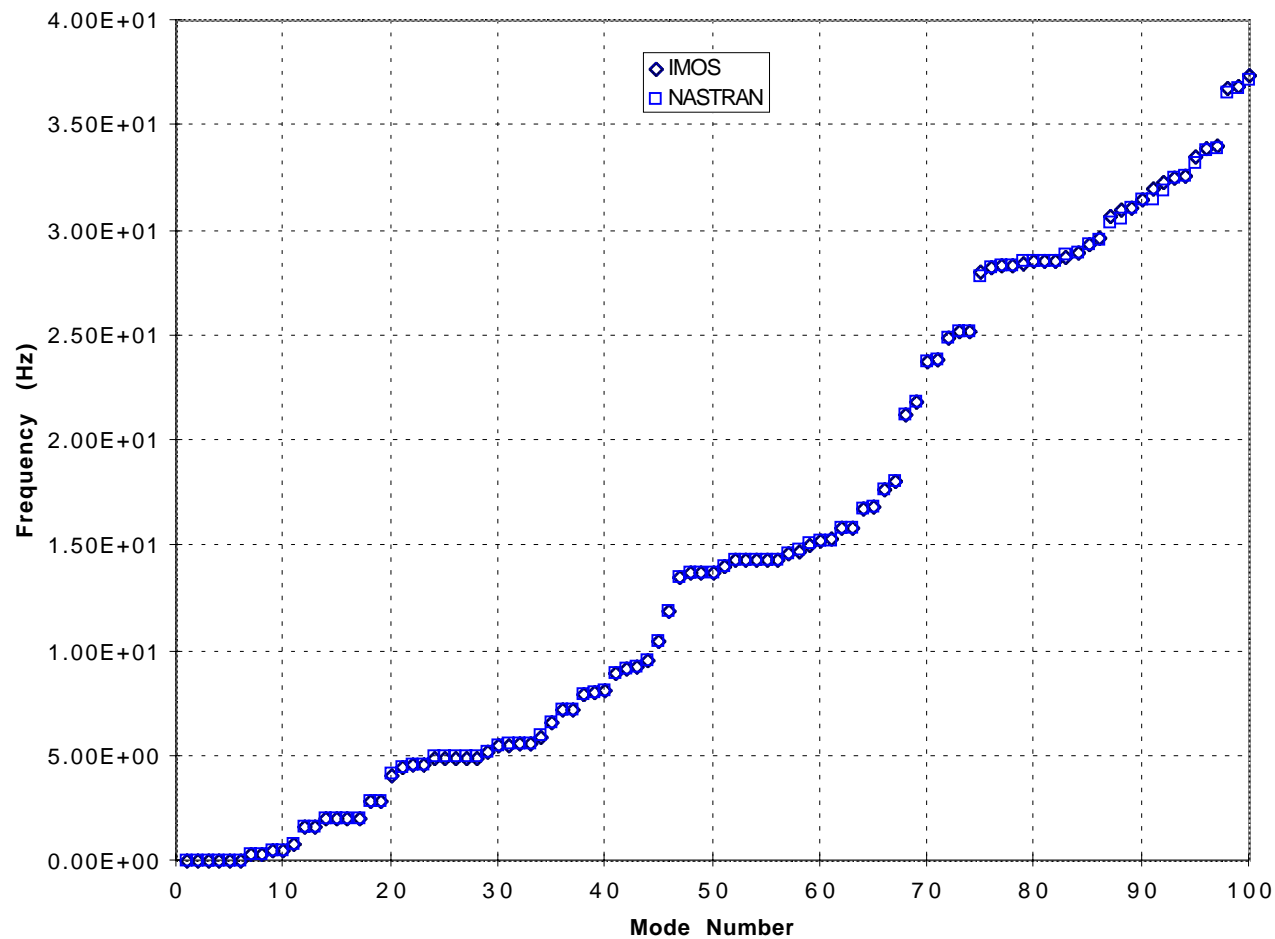
- # **Rigid Body Mode Strain Energy Should be Small**
  - automatic printout using rbmodesk
  - for NGST the largest component was found to be  $< 1.0e-7$
- # **Mass Properties Agree with NASTRAN**
  - using wtcgk, cg\_calck
  - Total mass= 2167 kg
  - Inertia Matrix about cg
  - Center of Gravity [xcg,ycg,zcg]= [-1.7050, -.02746, 1.4915]
- # **Normal Modes Analysis**
  - free-free boundary conditions (using speig with shifted k)
  - frequencies agree with NASTRAN (within 1%, see plot)
  - cross orthogonality of IMOS and NASTRAN modeshapes was good
    - ideally  $\Phi_{IMOS}^T * M_{IMOS} * \Phi_{NASTRAN} = I$
    - found max diff of .01 on diagonal, max .10 off-diag ( ave .0005 off-diag)
- # **Transient Dynamic Analysis Response Agrees with NASTRAN (see plot)**



# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Comparison of Modal Frequencies for NASTRAN and IMOS Models (v.6/97)



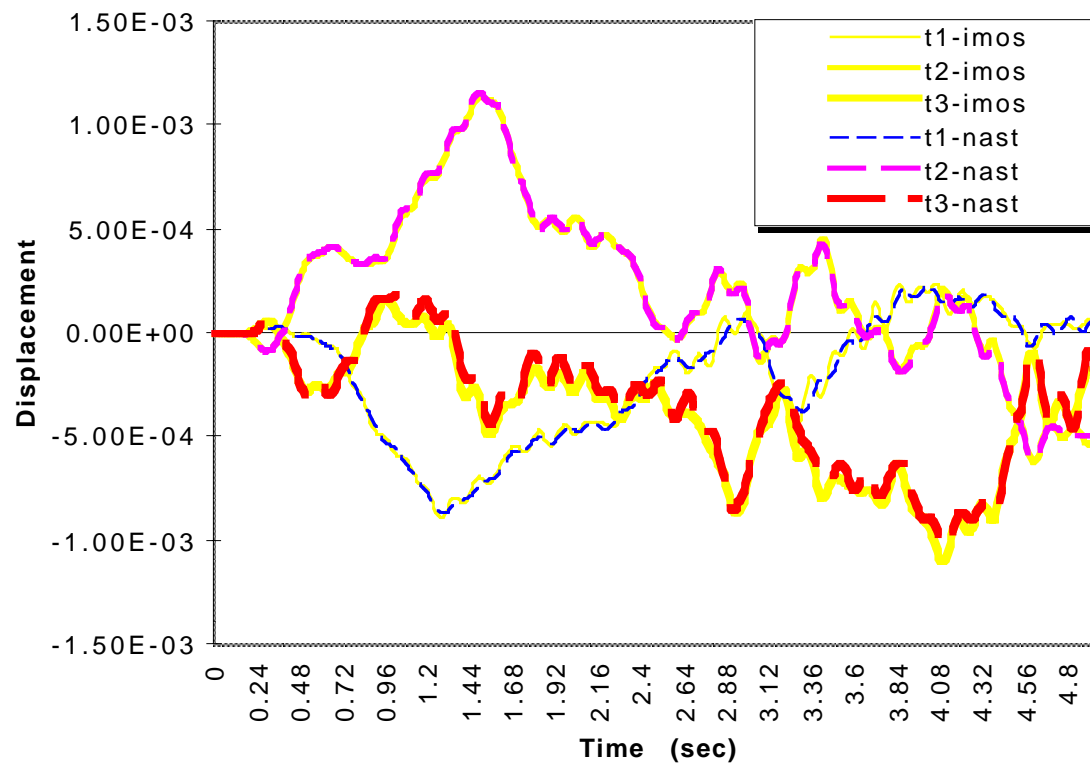




# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Comparison of NASTRAN and IMOS Transient Response Histories





# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Thermal Distortion Analysis

- # **Computed Mirror Distortion Vectors for Applied Temperature Profiles**
  - Temperatures provided by MSFC in NASTRAN bulk data format
    - converted to IMOS automatically with nas2imos
  - Steady State Ground to Orbit (used 100 deg K ref)
    - used cte's integrated over temp range
    - subtract ref temp to get delta T
  - Thermal Transient, 1 hr slew with 27 hr hold (43 time steps)
    - no dynamics involved- static response only
    - temps at T=0 used as reference
  - used tplate and tbeam (see viewgraph)
- # **Viewed Thermal Distortion using modelplot (static) and movmak (animation)**
- # **Compared Thermal Distortions from IMOS with NASTRAN**
  - difference is much less than 1%, (  $\text{norm}(\mathbf{u}_{\text{IMOS}} - \mathbf{u}_{\text{NASTRAN}}) / \text{norm}(\mathbf{u}_{\text{NASTRAN}})$  )
- # **Thermal Distortion Vectors Transmitted for Optical Performance Analysis**



# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS

## Script for Thermal Distortion Analysis



```
% This is a script to compute the displacement vector due to
% a temperature distribution (T) or gta.

niplate=[niquad; nitria];

ftoplate=tplate(niplate,xyz,propplt,mat,[1:length(niplate(:,1))]' );
% ftobar=tbeam(nibar,xyz,propbar,[1:length(nibar(:,1))]' );
ftobar=tbeam(nibar,xyz,propbar,[1:length(xyz(:,1))]' );

% Use Element temperatures
% etplate=tempvec(niplate,xyz,gta,0.0);
% [etbar,nt]=tempvec(nibar,xyz,gta,0.0);

% pgb=ftoplate*etplate + ftobar*etbar;
% pgb=ftoplate*etplate + ftobar*nt;

% Use Nodal temperatures- tn2e transforms nodal to elem temps
tn2e=tmode2el(niplate,xyz);
fto=ftoplate*tn2e + ftobar;
pgb=fto*T;

% transform thermal equiv forces from basic to local
t=gentloc(ti,tf);
pgl=t*pgb;
% reduce to nset and fset
if exist('gm')
    pgl(nset,:)=pgl(nset,:) + gm'*pgl(mset,:);
end
pfl=pgl(fset,:);

% compute displacements
phitl=k\pfl;
%phitl=pinv(full(k))*pfl;

% expand to gset
[np,mp]=size(phitl);
phigl=zeros(max(size(xyz))*6,mp);
phigl(fset,:)=phitl;
if exist('gm')
    phigl(mset,:)=gm*phigl(nset,:);
end

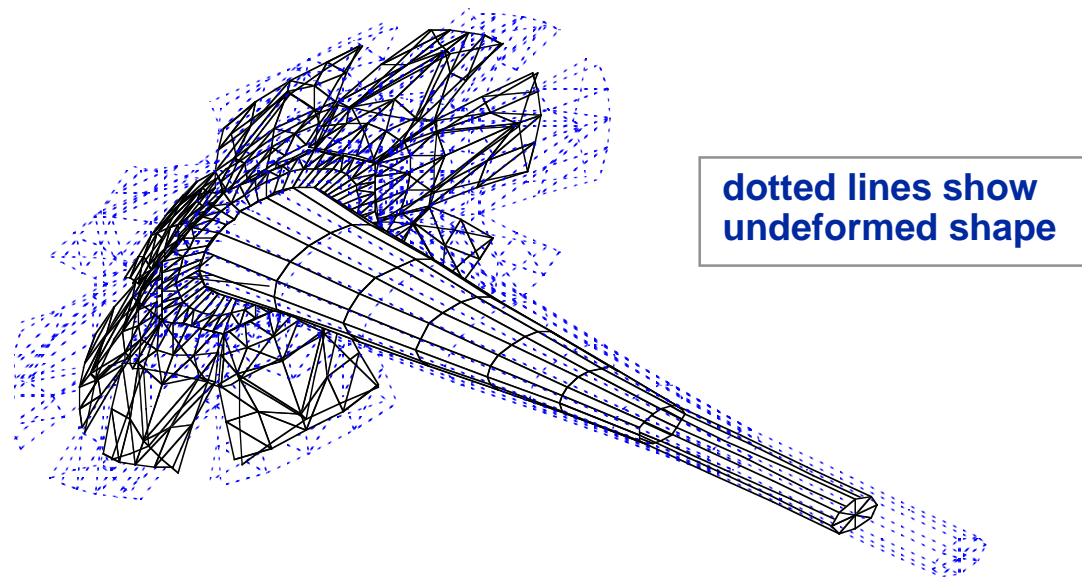
% transform to basic
phigb=t'*phigl;
```



# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Plot of Thermal Distortion Steady State, Ground to Orbit





## **NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS**



### **Using Substructuring for Very Large Models**

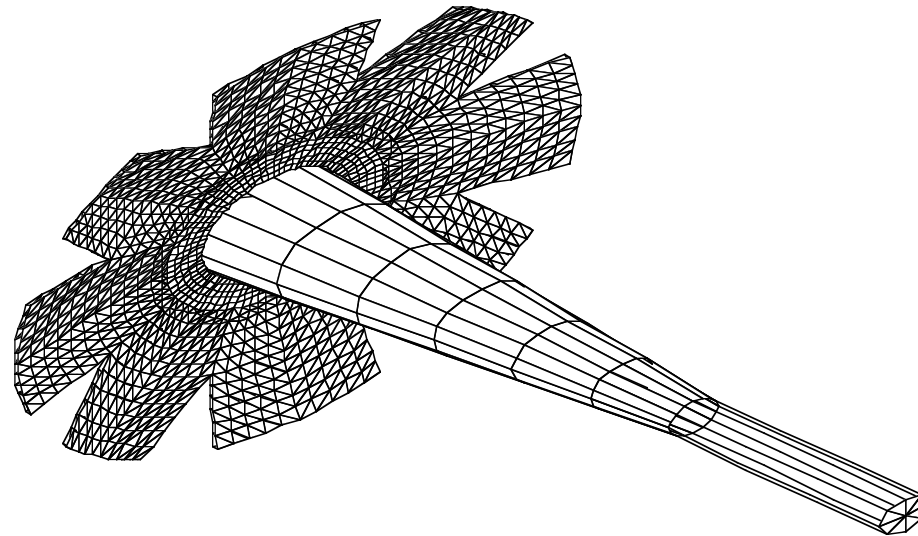
- # **NGST Model with Deformable Glass Primary Mirror has 20,550 dofs**
  - **Fine mesh required to capture actuator/structure interaction (448 actuators)**
    - **36 actuators per petal ( 8 petals)**
    - **160 actuators for central segment**
  - **Primary mirror model was already simplified**
    - **actuator spacing increased**
- # **“Out of memory “ when trying to solve large eigenproblem**
- # **Substructuring offered solution**



# **NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS**



## **NGST Telescope Model with Glass Primary (v.10/97)**





# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Using Substructuring for Very Large Models

- # Implemented Substructuring- using Craig Bampton method
  - divided the structure into 6 parts, taking advantage of symmetry
    - central segment pie section (192 boundary dofs)
      - found that whole central segment was too big- 6738 dofs
      - used 1/8 pie section- 924 dofs for eigenproblem
    - petal type 1 (24 boundary dofs) (1584 dofs for eigenproblem)
    - petal type 2 (24 boundary dofs) (1584 dofs for eigenproblem)
    - secondary mirror & support structure (96 boundary dofs)
    - science instrumentation assembly (6 boundary dofs)
    - isolation truss, bus/prop module & sunshield (9 boundary dofs)
  - used `cms_comp_cb.m` to perform analysis on each of the 6 parts (see viewgraph)
    - compute fixed interface normal modes (eigenvalue analysis)
    - compute constraint modes (static)



# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Using Substructuring for Very Large Models

# **Script to generate component mode data for petal type 1 substructure**

```
echo on
% define internal i-set and boundary b-set grid groups
ig1=[561:692,2177:2241,2243:2308];
ig1=[ig1' ones(length(ig1),6)];
bg1=[2097 1 1 1 0 0 0
     2102 1 1 1 1 1 1
     2107 1 1 1 0 0 0
     2242 1 1 1 1 1 1
     3166 1 1 1 1 1 1];

% get active internal and boundary dofs
is1=getdofs(ig1,xyz,bc);
bs1=getdofs(bg1,xyz,bc);

[knn1,kbb1,mnb1,mbb1,p1]=cms_comp_cb(k,m,is1,bs1,100);

% expand p to include dependent dofs, exclude spcs
[p1]=cms_comp_i2d(xyz,p1,ig1,is1,bs1,fset,spcgid,nset,mset,gm);

save petal_1c_cms knn1 kbb1 mnb1 mbb1 p1
```





# **NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS**



## **Using Substructuring for Very Large Models**

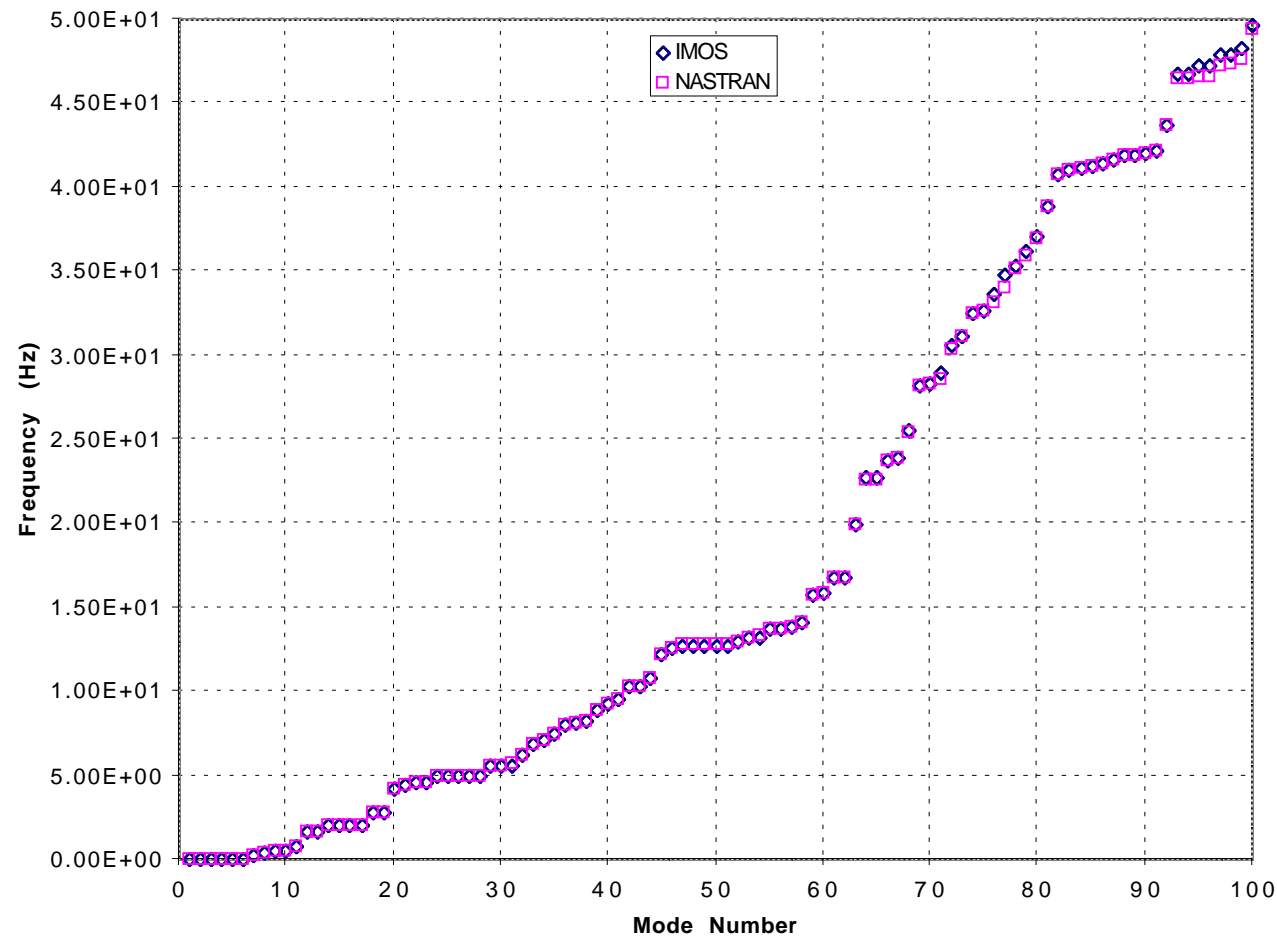
- # **Implemented Substructuring- using Craig Bampton method (cont'd)**
  - replicated central segment pie section 8 times
  - replicated petal types 1 & 2 - 4 times each
  - total of 19 components assembled (including replications)
  - resulting system eigenproblem size of 2715 dofs
- # **Comparison of resulting system frequencies between NASTRAN and IMOS**
  - frequencies match within 1% (see plot)



# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Comparison of Modal Frequencies for NASTRAN and IMOS Models (Glass v. 10/97) (IMOS model used Substructuring)





# NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS



## Summary of IMOS Experience with NGST

### # Eigenvalue Analysis

- speig sparse eigenvalue solver is slow for large problems
  - compiled version would help greatly
  - routine specializing in real-symmetric matrices would help too
  - have had bad results with small problems (< 1k dofs)
  - good results for large problems
- need to check results
  - compute residuals from the modal equation
    - $\mathcal{R} = K\Phi - M\Phi\Lambda$
    - compare norm of each modal residual vector with that of  $K\Phi$ 
      - can recompute modes in problem frequency range
  - check orthogonality, i.e.  $\Phi^T M \Phi$ 
    - can re-orthogonalize using onormphi (Gram-Schmidt) or other
  - sometimes small imaginary parts show up
    - can take real part if imaginary part is small enough
    - re-orthogonalize using onormphi



## **NEXT GENERATION SPACE TELESCOPE STRUCTURAL MODELING & ANALYSIS**



# **Summary of IMOS Experience with NGST (contd)**

### **# Thermal Distortion Analysis**

- need to re-check model after every modification
  - uniform cte with constant delta T (common sense, but ...)

### **# Substructuring**

- tedious up front bookkeeping, but effective method
  - subsequent studies using mat changes would be fast
- checking results of system eigen-analysis is essential (see prev viewgraph)
  - had to zero in on problem frequency ranges with speig
- can use it for statics as well- just using constraint modes

### **# Overall**

- favorable comparisons between IMOS and NASTRAN
- flexibility of working in a MATLAB environment accelerates productivity
- better eigen-solver would be a big help